

Airworthiness Directive

AD No.: 2026-0080

Issued: 17 April 2026

Note: This Airworthiness Directive (AD) is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation.

This AD is issued in accordance with Regulation (EU) 748/2012, Part 21.A.3B. In accordance with Regulation (EU) 1321/2014 Annex I Part M.A.301, or Annex Vb Part ML.A.301, as applicable, the continuing airworthiness of an aircraft shall be ensured by accomplishing any applicable ADs. Consequently, no person may operate an aircraft to which an AD applies, except in accordance with the requirements of that AD, unless otherwise specified by the Agency [Regulation (EU) 1321/2014 Annex I Part M.A.303, or Annex Vb Part ML.A.303, as applicable] or agreed with the Authority of the State of Registry [Regulation (EU) 2018/1139, Article 71 exemption].

Design Approval Holder's Name:

AIRBUS HELICOPTERS DEUTSCHLAND GmbH

Type/Model designation(s):

EC135 and EC635 helicopters

Effective Date: 01 May 2026

TCDS Number(s): EASA.R.009

Foreign AD: Not applicable

Supersedure: None

ATA – Aircraft Flight Manual – Emergency, Malfunction and Normal Procedures – Amendment

Manufacturer(s):

Airbus Helicopters; Airbus Helicopters Deutschland GmbH (AHD), formerly Eurocopter Deutschland GmbH; Eurocopter España S.A.

Applicability:

EC135 P1, EC135 P2, EC135 P2+, EC135 P3, EC135 T1, EC135 T2, EC135 T2+, EC135 T3, EC635 P2+, EC635 T1 and EC635 T2+ helicopters, all variants, all serial numbers (s/n).

Definitions:

For the purpose of this AD, the following definitions apply:

The RFM TR: Airbus Helicopters (AH) Rotorcraft Flight Manual (RFM) Temporary Revisions (TR) as listed in Appendix 1 of this AD, as applicable.

The updated procedures: Updated emergency, malfunction and normal procedures as defined in the RFM TR.

Reason:

Following the accomplishment of laboratory test, it has been determined that in case of loss of oil from the main gearbox (MGB), the 'low pressure' caution and/or warning might not be activated.

This condition, if not corrected, could lead to failure of the MGB, possibly resulting in loss of control of the helicopter.

To temporarily address this potential unsafe condition, AHD developed the updated procedures, and published the RFM TR, as defined in this AD.

For the reason described above, this AD requires to implement the updated procedures.

This AD is considered to be an interim action and further AD action may follow.

Required Action(s) and Compliance Time(s):

Required as indicated by this AD, unless the action(s) required by this AD have been already accomplished:

Operational Procedure Amendment:

- (1) Within 14 days or 50 flight hours, whichever occurs first after the effective date of this AD, implement the updated procedures, as defined in this AD.

Acceptable Method of Compliance:

- (2) Amending the RFM of a helicopter by inserting a copy of the RFM TR is an acceptable method to comply with the requirements of paragraph (1) of this AD for that helicopter.
- (3) Amending the RFM of a helicopter by incorporating a later EASA approved RFM revision, which includes the RFM TR, is an acceptable method to comply with the requirements of paragraph (1) of this AD for that helicopter.

Ref. Publications:

AH RFM TR as listed in Appendix 1 of this AD.

The use of later approved revisions of the above-mentioned document is acceptable for compliance with the requirements of this AD.

Remarks:

1. If requested and appropriately substantiated, EASA can approve Alternative Methods of Compliance for this AD.
2. Based on the required actions and the compliance time, EASA have decided to issue a Final AD with Request for Comments, postponing the public consultation process until after publication. All interested persons may send their comments, referencing the AD Number, to the E-mail address specified in below Remark 3, prior to 15 May 2026. Only if any comment is received during the consultation period, a Comment Response Document will be published in the [EASA Safety Publications Tool](#), in a compressed ('zipped') file, attached to the record for this AD.
3. Enquiries regarding this AD should be referred to the EASA Safety Information Section, Certification Directorate. E-mail: ADs@easa.europa.eu.



4. Information about any failures, malfunctions, defects or other occurrences, which may be similar to the unsafe condition addressed by this AD, and which may occur, or have occurred on a product, part or appliance not affected by this AD, can be reported to the [EU aviation safety reporting system](#). This may include reporting on the same or similar components, other than those covered by the design to which this AD applies, if the same unsafe condition can exist or may develop on an aircraft with those components installed. Such components may be installed under an FAA Parts Manufacturer Approval (PMA), Supplemental Type Certificate (STC) or other modification.

5. For any question concerning the technical content of the requirements in this AD, please contact: Airbus Helicopters Deutschland GmbH, Industriestrasse 4, 86609 Donauwörth, Federal Republic of Germany;
Web portal: <https://airbusworld.helicopters.airbus.com/>
E-mail: customersupport.helicopters@airbus.com.



Appendix 1

- Rotorcraft Flight Manual FLM EC135 P1(CDS) Temporary Revision 24
- Rotorcraft Flight Manual FLM EC135 P1(CPDS) Temporary Revision 19
- Rotorcraft Flight Manual FLM EC135 P2(CPDS) Temporary Revision 17
- Rotorcraft Flight Manual FLM EC135 P2+ Temporary Revision 12
- Rotorcraft Flight Manual FLM EC135 P3(CPDS) Temporary Revision 06
- Rotorcraft Flight Manual FLM EC135 P3H Temporary Revision 04
- Rotorcraft Flight Manual FLM EC135 T1(CDS) Temporary Revision 27
- Rotorcraft Flight Manual FLM EC135 T1(CPDS) Temporary Revision 22
- Rotorcraft Flight Manual FLM EC135 T2(CPDS) Temporary Revision 18
- Rotorcraft Flight Manual FLM EC135 T2+ Temporary Revision 11
- Rotorcraft Flight Manual FLM EC135 T3(CPDS) Temporary Revision 05
- Rotorcraft Flight Manual FLM EC135 T3H Temporary Revision 06
- Rotorcraft Flight Manual FLM EC635 P2+ Temporary Revision 12
- Rotorcraft Flight Manual FLM EC635 T1(CPDS) Temporary Revision 22
- Rotorcraft Flight Manual FLM EC635 T2+ Temporary Revision 11
- Rotorcraft Flight Manual Supplement FMS EC135 P3H 9.1-4 Temporary Revision 04
- Rotorcraft Flight Manual Supplement FMS EC135 T3H 9.1-4 Temporary Revision 06

